

# A Remote Anomaly Detection System for Slocum Underwater Gliders

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## Abstract

Marine Autonomous Systems (MAS) operating at sea beyond visual line of sight need to be self-reliant, as any malfunction could lead to loss or pose a risk to other sea users. In the absence of fully automated on-board control and fault detection tools, MAS are piloted and monitored by experts, resulting in high operational costs and limiting the scale of observational fleets that can be deployed simultaneously. Hence, an effective anomaly detection system is fundamental to increase fleet capacity and reliability. In this study, an on-line, remote fault detection system is developed for underwater gliders. Two alternative methods are analysed using time series data: feed-forward deep neural networks estimating the glider's vertical velocity and an autoencoder. The systems are trained using field data from four baseline deployments of Slocum gliders and tested on six deployments of vehicles suffering from adverse behaviour. The methods are able to successfully detect a range of anomalies in the near real time data streams, whilst being able to generalise to different glider configurations. The autoencoder's error in reconstructing the original signals is the clearest indicator of anomalies. Thus, the autoencoder is a prime candidate to be included into an all-encompassing condition monitoring system for MAS.

*Keywords:* fault detection, anomaly detection, underwater gliders,  
autonomous underwater vehicles, autoencoder, deep learning, system  
identification

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13 **Nomenclature**

|    |                        |  |
|----|------------------------|--|
| 14 | $\alpha$               | angle of attack [°]  |
| 15 | $\alpha_T$             | glider thermal expansivity [°C <sup>-1</sup> ]                   |
| 16 | $\beta$                | glide-path angle [°]   |
| 17 | $\ddot{z}$             | vertical acceleration in the inertial frame [cm/s <sup>2</sup> ] |
| 18 | $\delta B$             | buoyancy offset [N]  |
| 19 | $\delta_r$             | rudder angle [°]   |
| 20 | $\dot{\delta}_r$       | rate of change of rudder angle [°/s]                             |
| 21 | $\dot{\phi}$           | roll angular velocity [°/s]                                      |
| 22 | $\dot{\psi}$           | yaw angular velocity [°/s]                                       |
| 23 | $\dot{\theta}$         | pitch angular velocity [°/s]                                     |
| 24 | $\dot{V}_{\text{vbd}}$ | rate of change of VBD volume [cm <sup>3</sup> /s]                |
| 25 | $\dot{x}_b$            | rate of change of battery position [cm/s]                        |
| 26 | $\dot{z}$              | vertical velocity in the inertial frame [cm/s]                   |
| 27 | $\epsilon_c$           | glider absolute compressibility [dbar <sup>-1</sup> ]            |
| 28 | $\phi$                 | roll angle [°]   |
| 29 | $\rho$                 | water density [kg/m <sup>3</sup> ]                               |
| 30 | $\theta$               | pitch angle [°]  |
| 31 | $\tilde{\psi}$         | yaw angle [°]  |
| 32 | $f_1$                  | engineered feature [kg]  |
| 33 | $f_2$                  | engineered feature [kg]  |
| 34 | $g$                    | gravitational acceleration [m/s <sup>2</sup> ]                   |
| 35 | $m$                    | glider flooded mass [kg]   |

|    |                  |  |
|----|------------------|--|
| 36 | $p$              | water pressure [dbar]                                    |
| 37 | $S$              | glider wetted surface area [m <sup>2</sup> ]             |
| 38 | $T$              | water temperature [°C]                                   |
| 39 | $T_0$            | reference temperature [°C]                               |
| 40 | $V_0$            | glider reference volume [m <sup>3</sup> ]                |
| 41 | $V_{\text{vbd}}$ | Variable Buoyancy Device (VBD) volume [cm <sup>3</sup> ] |
| 42 | $x_b$            | battery position [cm]                                    |
| 43 | $z$              | vertical position in the inertial frame [m]              |

## 44 **1. Introduction**

### 45 *1.1. Background and Motivation*

46 The pervasive adoption of Marine Autonomous Systems (MAS) is cur-  
 47 rently constrained by the challenges of operating fully independently of sup-  
 48 port vessels, in the demanding dynamic environment of the world’s oceans.  
 49 MAS are piloted and monitored by experts, keeping operational costs high  
 50 and limiting the scale of observational fleets that can be deployed simultane-  
 51 ously (Verma and Simmons, 2006). Since a MAS can be at sea for months at  
 52 a time, operating ‘over-the-horizon’ from human pilots and support vessels, it  
 53 is impossible to perform reactive maintenance, i.e. maintenance when a fault  
 54 occurs. Therefore, strategies must be developed for automated fault diagnos-  
 55 tics and prognostics in a predictive maintenance framework, i.e. identifying  
 56 when maintenance should be preemptively scheduled on individual compo-  
 57 nents, by continuously monitoring conditions and MAS behaviour during  
 58 operation. Current manual detection, diagnosis and mitigation of problems  
 59 are limited by the experience of the individual pilot and are subject to human  
 60 error, especially when a MAS platform requires pilot attention around the  
 61 clock.

62 In the absence of general on-board anomaly detection and diagnosis sys-  
 63 tems, the ability to transmit sensor data in a timely manner to an off-board  
 64 system or human operator and to receive appropriate commands in response  
 65 becomes of critical importance for MAS safety and performance.

66 Anomaly detection is currently extremely challenging for Autonomous  
67 Underwater Vehicles (AUVs), as they are under-observed systems, with lim-  
68 ited or decimated data available via satellite for pilots to interpret when the  
69 vehicle comes periodically to the surface. If the underlying cause of observed  
70 adverse behaviour cannot be correctly diagnosed and the situation remedied  
71 (e.g. via the remote adjustment of piloting parameters or mission scope),  
72 the vehicle, their cargo or data can be lost or present a hazard to shipping  
73 (Thieme and Utne, 2017).

#### 74 *1.2. Literature Review*

75 Underwater Gliders (UGs) are a type of AUV that are being used ex-  
76 tensively for long-term observation of key physical oceanographic parameters  
77 (Rudnick, 2016). The vehicles achieve vertical motion in the water column by  
78 changing their buoyancy through a variable buoyancy device (VBD). Wings  
79 generate a forward motion component from the vertical motion. Hence, UGs  
80 travel in a characteristic sawtooth pattern in the vertical plane. Their simple  
81 propulsion system, which consists of the VBD, pitch control and either roll  
82 control or a rudder, is highly efficient. Therefore, although they operate at  
83 a low surge speed ( $\approx 0.3$  m/s), the deployments of UGs can last for sev-  
84 eral months. Reviews of the initial glider technologies can be found in Davis  
85 et al. (2003) and Wood (2009), whilst Rudnick (2016) and Testor et al. (2019)  
86 contain good summaries of their applications in oceanography.

87 The reliability of UGs is analysed thoroughly in Brito et al. (2014), with  
88 the authors collecting failure data from most European operators. The most  
89 common failures have been observed to be leakages, failures of electrical (e.g.  
90 the battery, satellite communication hardware), mechanical components (e.g.  
91 the VBD pump and bladder, rudder and roll motor), navigation and scien-  
92 tific sensors, and software errors. Leaks, motor or pump malfunctions, low-  
93 battery voltage and sensor drop-outs or faults are also identified as problems  
94 in Schofield et al. (2007) Additionally, Frajka-Williams et al. (2011) have  
95 observed the failure of the pitch tilt sensor. Furthermore, the authors have  
96 investigated the loss of wing for UGs in Anderlini et al. (2020a) and the  
97 impact of biofouling in Anderlini et al. (2020b).

98 Methods for fault detection and diagnostics can be subdivided into rule-  
99 based, model-based and data-driven solutions (Hamilton et al., 2007).

100 Rule-based methods rely on bespoke heuristics, usually in the form of if-  
101 then statements, obtained from designers' observations of the system. Model-  
102 based approaches rely on dynamic models of the physical systems. Hence,

103 model-based methods are highly suitable for condition monitoring of new  
104 systems where available data is limited. Model-based techniques are robust,  
105 simple and relatively inexpensive computationally, but multiple systems are  
106 usually run concurrently for redundancy. Specifically for AUVs, Freddi et al.  
107 (2013) design a system to detect faulty thrusters. Hamilton et al. (2007)  
108 propose an integrated fault detection and diagnosis architecture for AUVs,  
109 although the focus is on on-board systems.

110 Data-driven solutions rely on the analysis of actual sensor data, thus  
111 showing significant improvements in accuracy when large data sets are avail-  
112 able. In particular, deep-learning-based methods can be generalised to dif-  
113 ferent applications and can be scaled to a large number of sensors, but need  
114 many samples for training.

115 Data-driven condition monitoring solutions, with a focus on machine and  
116 deep learning, are surveyed in Ellefsen et al. (2019) for autonomous ships,  
117 with (Fink et al., 2020) providing an overview of present and current trends  
118 in data-driven fault diagnostics and prognostics. Specifically for AUVs, data-  
119 driven fault detection methods include radial basis function networks (Wang  
120 and Zhang, 2006), Gaussian particle filter (Sun et al., 2016) and artificial  
121 immune system (Yao et al., 2018). Raanan et al. (2018) have introduced  
122 an automatic fault detection system for long-range AUVs based on Bayesian  
123 nonparametric topic modelling techniques. Although the data set focuses on  
124 the identification of bottoming events, the behaviour of the analysed long-  
125 range AUV is similar to that of UGs. A system to develop safety indicators  
126 for the operation of MAS is described in Thieme and Utne (2017), with a  
127 case study on an AUV. [An anomaly detection system for UGs specific to  
128 marine biofouling was presented in Anderlini et al. \(2021\) using ensembles  
129 of regression trees and K-means clustering.](#)

130 Modern deep learning solution specific to anomaly detection are sum-  
131 marised in Pang et al. (2021). Of the three main deep anomaly detection  
132 paradigms, learning the feature representations of normality is of interest  
133 for the UG application, i.e. being able to differentiate between normal and  
134 abnormal operating conditions with anomaly scoring. These methods can  
135 be further subdivided into generic feature learning and anomaly measure-  
136 dependent feature learning. Whereas the latter aims at learning feature  
137 representations that are optimised for one specific existing anomaly mea-  
138 sure, the former is more general and can deal well with imbalanced datasets  
139 heavily skewed towards normal operating conditions due to the low anomaly  
140 detection recall rate. Autoencoders are a very powerful, but simple strat-

141 egy that consists of a network that learns to reconstruct the original signals  
142 passing through a smaller latent space that summarises only the fundamen-  
143 tal information removing the effect of noise. Their generality makes them  
144 well suited for the application to UGs. An alternative modification specific  
145 to UGs consists of reconstructing only the vertical velocity signal based on  
146 the other sensors. This procedure is similar to the model-based anomaly  
147 detection strategies (Anderlini et al., 2020a), but with a model with many  
148 more parameters. In particular, the difference between a machine learning  
149 implementation that relies on engineered features from the dynamic model  
150 and a deep learning strategy that relies only on the raw variables should be  
151 investigated.

### 152 *1.3. Contribution*

153 The operation of MAS platforms beyond visual line of sight requires  
154 a suitable command and control system. For example, the UK National  
155 Oceanography Centre are designing a new command-and-control system for  
156 efficient MAS fleet management (Farley et al., 2019; Harris et al., 2020) to fa-  
157 cilitate the over-the-horizon operation of their ever-increasing fleet of AUVs.  
158 Another example is the LSTS Neptus and Dune over-the-horizon command-  
159 and-control environment (Sousa Dias et al., 2005; Madureira et al., 2013;  
160 Pinto et al., 2013) Control and command systems will require an effective  
161 condition monitoring tool to enable round-the-clock operations. Rule-based  
162 and model-based over-the-horizon fault detection methods have been intro-  
163 duced for UGs in Anderlini et al. (2020a) and Anderlini et al. (2020b) for  
164 specific faults, namely wing loss and marine growth. However, more general  
165 solutions are required.

166 This work introduces the first data-driven fault detection system for UGs.  
167 In particular, two alternative methods are investigated: an autoencoder ex-  
168 tracting useful features in a latent space, and a deep neural network (DNN)  
169 regression approach estimating the UG’s vertical velocity. The systems are  
170 trained with the steady-state flight data from four deployments of healthy  
171 Slocum UGs and tested on six deployments of Slocum UGs suffering from  
172 adverse behaviour. The performance of the data-driven methods is assessed  
173 against existing rule- and model-based solutions (Anderlini et al., 2020a,b)  
174 to show the improvement in generality.

175 *1.4. Article Organisation*

176 In Section 2.1, the operations of Slocum UGs and the analysed datasets  
177 are described. The novel data-driven anomaly detection methods are intro-  
178 duced in Section 3. Then, the results are presented and discussed in Section 4,  
179 with concluding remarks being made in Section 5.

180 **2. Slocum Gliders Data**

181 *2.1. Slocum Underwater Gliders*

182 There are range of manufacturers of underwater gliders (Rudnick, 2016).  
183 This study is focused on 200m depth rated Slocum G2’s from Teledyne Webb  
184 Research (Webb et al., 2001; Schofield et al., 2007). An example can be seen  
185 in Fig. 1.

186 As shown in Fig. 2, the Slocums are actuated by a VBD, which consists in  
187 an oil bladder that can be extended or retracted from the pressure hull. When  
188 the bladder is outside the pressure hull, the vehicle’s displacement increases  
189 and so does its net buoyancy and vice versa, resulting in their characteristic  
190 ‘yo-yo’ motion. The vehicles considered in this study are limited to changes  
191 in the VBD volume of  $\pm 250 \text{ cm}^3$ . Furthermore, pitch is controlled by shifting  
192 the position of one movable battery pack with a dedicated mechanism. Yaw  
193 is controlled via a rudder, which is magnetically coupled to a servo motor to  
194 avoid an opening in the pressure hull.

195 Example time series data for a typical dive cycle can be seen in Fig. 3.  
196  $z$  indicates the vertical position of the vehicle in the water column (posi-  
197 tive upwards), which is measured by an on-board pressure sensor. Its time-  
198 derivative yields  $\dot{z}$ , the vertical velocity. The actuators’ control signals are  
199 the volume of the VBD,  $V_{\text{vbd}}$ , the position of the moving battery pack,  $x_{\text{b}}$ ,  
200 and the rudder angle,  $\delta_{\text{r}}$ . The roll,  $\phi$ , and pitch,  $\theta$ , angles are measured by  
201 tilt sensors, while a magnetic compass indicates the heading angle,  $\psi$ . In  
202 this article, the difference of the instantaneous yaw angle and the mean yaw  
203 angle over the whole dive cycle,  $\bar{\psi}$ , is used to favour the body-fixed over the  
204 inertial reference frame.

205 As shown in Fig. 2 and Fig. 3a, Slocums can perform multiple ‘yos’ per  
206 dive (two in this particular case). This means that the vehicle can sample  
207 the water column multiple times before returning to the surface to send and  
208 receive data by satellite and get a new position fix via GPS (Teledyne Webb  
209 Research, 2012). The data sent ashore needs to be decimated to reduce the  
210 time that the vehicle spends on the surface to lower a) the risk of damage



Figure 1: Slocum G2 operated by the NOC.

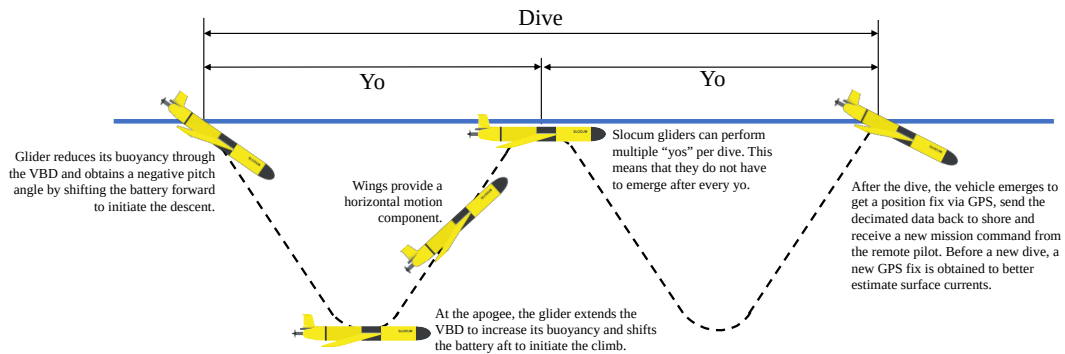


Figure 2: Diagram showing the concept of operation of a Slocum UG. The drawing is not to scale: the analysed vehicles reach their apogee at 200-m-depth and have glide path angles with a magnitude in the range  $(15^\circ, 30^\circ)$ .

211 from collisions with other sea users and wave loads, b) the power expenditure  
 212 and c) the actual financial cost associated with the transmission of the data  
 213 by satellite (specifically, via Iridium). The data usually includes the vehicle's  
 214 orientation, its depth (from which the vertical velocity can be obtained),  
 215 the actuator's signals, the capacity and voltage of the battery, the estimated  
 216 position, samples of the scientific data of interest and warnings from on-board  
 217 health monitoring systems. The decimation means that the samples have a



218 low rate, typically with a time step of 30-60 s for signals that are considered  
219 of least importance (e.g. the roll angle) and 10 s for the signals that are of  
220 most interest, e.g. the scientific data. In the analysed data sets, each dive  
221 consisted of up to 14 yos and lasted up to four hours.

222 Fig. 3b, Fig. 3c and Fig. 3e show the control input and output for the  
223 VBD, battery linear actuator and rudder. From a comparison with Ander-  
224 lini et al. (2019b), it is possible to notice that the rudder enables Slocums  
225 to exhibit a much smoother response than other UG technologies, even for  
226 shallower dives where disturbances are stronger. In Fig. 3c, it is also inter-  
227 esting to note that the direction of travel of the battery is opposite to the  
228 pitch angle, which is due to the actuator’s reference frame.

229 To control UGs, pilots rely on the surface dialogue, with a user-friendly  
230 interface providing information on the vehicle’s health status, current mission  
231 plan, last GPS position and the decimated data from past dives. Typically,  
232 during normal operation, the remote pilots will first check the surface dia-  
233 logue for errors, warning and oddities from the glider, along with the dive  
234 profile to ensure it is symmetrical and the glider is reaching the target depth.  
235 The most common errors identified on-board are relatively mild aborts, e.g.  
236 glider stalls, behaviour errors, and communication interruptions (Schofield  
237 et al., 2007). Progress towards the target waypoint is also considered, along  
238 with a check of the battery health and consumption. This full check is usu-  
239 ally performed once per day, with the pilot making smaller observations more  
240 regularly after each dive. Therefore, pilots are only likely to look into the  
241 flight parameters in detail if the glider is reporting errors, is failing to dive  
242 correctly or is not making progress towards a waypoint. Hence, issues with  
243 roll, for instance, can go unnoticed.

## 244 *2.2. Dataset Description*

245 This study uses the data measured by Slocum G2 gliders over ten deploy-  
246 ments operated by the NOC (BODC, 2021). A summary of the missions,  
247 including unit (or UG) ID, mission date and location, can be found in Ta-  
248 ble 1. As can be seen, the vehicles were operated in either the Celtic or North  
249 Seas between 2014 and 2019.

250 The missions in the Celtic Sea were run as part of project AtlantOS  
251 (EC/633211), with the work extended and complemented by project CaN-  
252 DyFloSS: ‘Carbon and Nutrient Dynamics and Fluxes over Shelf Systems’  
253 (NE/K001701/1). The project relied on multiple deployments of UGs to test

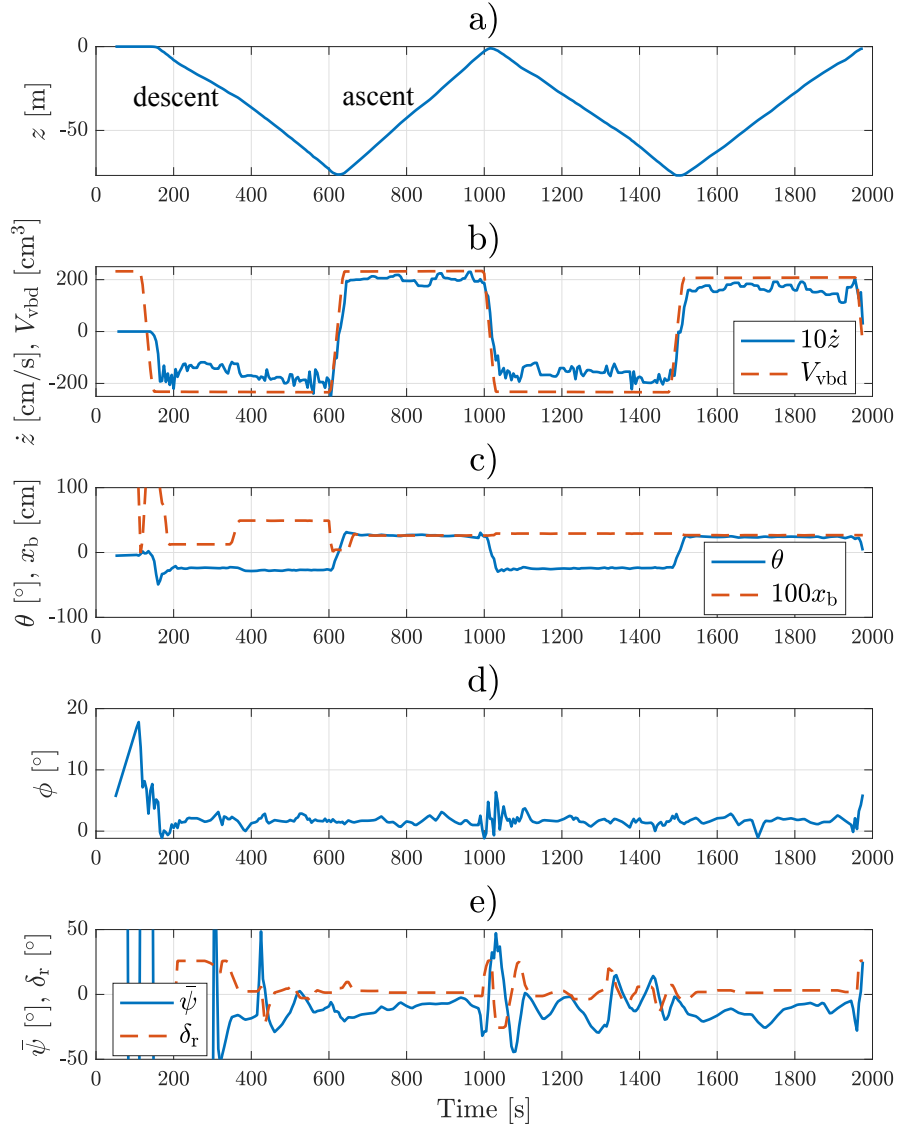


Figure 3: Example dive cycle of an intact Slocum glider.

254 the capability of MAS to synoptically assess physical and biogeochemical  
 255 functioning in shelf sea systems (Palmer et al., 2018).

256 The missions in the North Sea were part of project ALTERECO: "An Al-  
 257 ternative Framework to Assess Marine Ecosystem in Shelf Seas" (NE/P013902/2).  
 258 In this project, the UGs were used to validate a novel monitoring framework

Table 1: Summary of the analysed deployments (BODC, 2021), including the hydrostatics of the UGs.

| No. | Unit | Date | Location   | Duration [days] | $m$ [kg] | $V_0$ [cm <sup>3</sup> ] | $T_0$ [°C] |
|-----|------|------|------------|-----------------|----------|--------------------------|------------|
| 1   | 345  | 2014 | Celtic Sea | 123.9           | 58.863   | 57398.8                  | 19.98      |
| 2   | 345  | 2019 | North Sea  | 76.8            | 64.587   | 63017.6                  | 19.64      |
| 3   | 397  | 2015 | Celtic Sea | 45.9            | 59.400   | 57990.4                  | 15.90      |
| 4   | 419  | 2015 | Celtic Sea | 11.0            | 65.236   | 63682.0                  | 15.88      |
| 5   | 194  | 2017 | North Sea  | 83.9            | 58.631   | 57226.0                  | 20.83      |
| 6   | 399  | 2015 | Celtic Sea | 84.6            | 65.258   | 63665.8                  | 16.80      |
| 7   | 423  | 2015 | Celtic Sea | 6.8             | 65.850   | 64303.2                  | 16.13      |
| 8   | 424  | 2015 | Celtic Sea | 20.8            | 66.026   | 64435.3                  | 16.13      |
| 9   | 304  | 2019 | North Sea  | 76.9            | 59.044   | 57615.8                  | 19.23      |
| 10  | 436  | 2019 | North Sea  | 89.8            | 65.281   | 63716.5                  | 19.45      |

259 to deliver improved understanding in time and space shelf sea ecosystem  
 260 health and functioning (Matthew R. Palmer et al., 2020).

261 Gliders were deployed and recovered from a mixture of large and small  
 262 research vessels as well as fishing boats. Once deployed, gliders undertook  
 263 sustained observations for multi-month periods, undertaking repeated tran-  
 264 sects in the relevant operating area. The only human intervention was via  
 265 remote pilots, up until the point of recovery. By cycling the gliders sustained  
 266 observations can be extended to multiple years.

### 267 3. Anomaly Detection

268 Here, a novel data-driven anomaly detection system for UGs is developed  
 269 to identify adverse behaviour over the horizon, as summarised schematically  
 270 in Fig. 4. The method is developed and tested off-line using the deployment  
 271 data sets shown in Table 1. The system is trained using baseline data from  
 272 deployments where the vehicle exhibited healthy status and tested with de-  
 273 ployments datasets exhibiting adverse vehicle behaviour. Once the efficacy  
 274 of the system is established, the trained anomaly detection schemes can be  
 275 used on-line from the remote control centre to notify pilots of possible faults  
 276 with the UG after each surfacing and satellite connection, as shown in Fig. 2.  
 277 It is important to note that vehicles store on-board all data samples, whilst  
 278 they send by satellite only decimated data due to the associated cost and  
 279 power loss. Hence, the training data can be subjected to deeper cleaning  
 280 than the decimated real-time data.

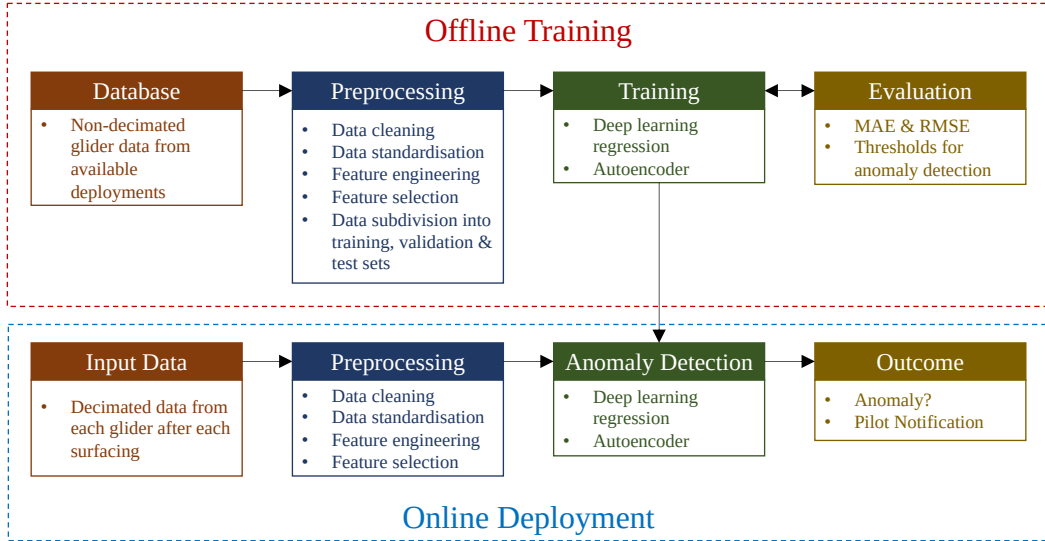


Figure 4: Diagram of the new fault detection system for Slocum underwater gliders.

281 Two alternative data-driven solutions are investigated: an autoencoder  
 282 extracting useful features in a latent space, and a deep neural network (DNN)  
 283 regression approach estimating the UG’s vertical velocity. In the following  
 284 sections, the preprocessing step and the development of the individual fault  
 285 detection schemes are treated in detail.

### 286 3.1. Steady-State Flight Model of a UG

287 As described in Sec. 2.1, UGs operate in steady-state conditions for most  
 288 of the descent and ascent of each yo. The free-body diagram of the equilib-  
 289 rium condition for the steady-state flight is shown schematically in Fig. 5a  
 290 and Fig. 5b for descents and ascents, respectively.  $B$  indicates the net buoy-  
 291 ancy,  $L$  the lift and  $D$  the drag force.  $U$  is the surge velocity component in  
 292 the body-fixed frame,  $\theta$  the pitch,  $\alpha$  the attack and  $\beta$  the glide-path angles.  
 293 The glide-path angle indicates the angle of the flight path in the inertial refer-  
 294 ence system and is obtained from the sum of the pitch and attack angles:  
 295  $\beta = \theta + \alpha$ .

296 In both descents and ascents, the balance of forces in equilibrium resulting  
 297 in steady state flight yields (Merckelbach et al., 2019)

$$B - L \cos \beta - D \sin \beta = 0. \quad (1)$$

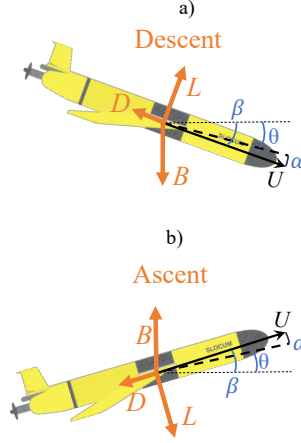


Figure 5: Free-body diagram of the intact glider in dives (a) and climbs (b) (profile view).

298 This persists over the majority of a dive cycle. The drag and lift forces can  
 299 be modelled as

$$L = \frac{1}{2} k_L \alpha \rho S U^2, \quad (2)$$

300

$$D = \frac{1}{2} (k_{D,0} + k_{D,L} \alpha^2) \rho S U^2, \quad (3)$$

301 where  $\rho$  is the water density,  $S$  is the wetted surface area,  $k_L$ ,  $k_{D,0}$  and  $k_{D,L}$   
 302 are constants used to compute the lift, drag and induced drag coefficients.  
 303 The water density is obtained from the water pressure, salinity and tem-  
 304 perature (and the derived conductivity) using the Gibbs Seawater Toolbox  
 305 (McDougall and Barker, 2011). These properties are measured by a Conduc-  
 306 tivity, Temperature and Density (CTD) sensor.

307 The net buoyancy force can be computed as

$$B = g \{-m + \rho [V_0 (1 - \epsilon_c p + \alpha_T (T - T_0)) + V_{\text{vbd}}]\} + \delta B, \quad (4)$$

308 where  $g$  is the gravitational acceleration,  $m$  is the UG mass,  $V_0$  its reference  
 309 volume,  $\epsilon_c$  the absolute compressibility of the pressure hull and  $\alpha_T$  its thermal  
 310 expansivity, with the reference temperature  $T_0$ .  $p$  is the water pressure and  $T$   
 311 its temperature. The offset in buoyancy  $\delta B$  is added to account for possible  
 312 changes in the net buoyancy of the vehicle after faults.

## 313 3.2. Preprocessing

### 314 3.2.1. Data Cleaning

315 As UGs are in steady-state flight for most of their deployment, the devel-  
316 opment of data-driven anomaly detection strategies can be greatly simplified.  
317 However, data cleaning is necessary to discard transient effects. In this study,  
318 the recovery-mode data stored on-board the gliders in the .DBD and .EBD  
319 files is used for the training and evaluation of the fault detection schemes (Tele-  
320 dyne Webb Research, 2012). The data is converted from binary to ASCII  
321 format using the Python *dbdreader* module<sup>1</sup>. Subsequently, the points are  
322 imported into the MATLAB environment.

323 Due to the large size of the available datasets (more than 8 million points  
324 for all combined deployments), all dive cycles for which any of the signals  
325 of interest are unavailable are removed. Additionally, a time vector starting  
326 from 0 s is created for every cycle. In a further cleaning sweep, all cycles with  
327 either a maximum depth shallower than 25 m or less than 10 time stamps  
328 are removed.

329 The greatest challenge for the cleaning process is represented by the time  
330 synchronisation: all sensors sample at slightly different time stamps. Fur-  
331 thermore, the navigation and scientific (for the CTD) Central Processing  
332 Units (CPUs) are synced only at the surface, so that a time drift is notice-  
333 able (with a mean drift of 3 s over 4 hr). Hence, all signals are resampled  
334 by linear interpolation for exactly the same time stamps, with a time step of  
335 5 s. The navigation and scientific computers are synced through the pressure  
336 signal, which is measured by both units.

337 Afterwards, the variables of interest, such as vertical velocity and water  
338 density, are computed from the raw signals. Furthermore, the signals are  
339 modified to reflect the units shown in the nomenclature. Additionally, the  
340 data is split into individual yo and their dive and climb stages. In Fig. 3b,  
341 the vertical velocity signal is noisier than the VBD volume signal. Hence,  
342 changes in sign of the difference between values of  $V_{\text{vbd}}$  for neighbouring time  
343 stamps are used to identify individual dives and climbs.

344 To reduce the transient effects caused by operations on the surface and  
345 apogee, points within 15 m from the surface (or the highest point of the  
346 climb) and the maximum depth are ignored. Furthermore, only steady-state  
347 data are kept by removing points which present significant changes in the

---

<sup>1</sup><https://pypi.org/project/dbdreader/>

Table 2: Upper limits used in the data cleaning.

| Variable                    | Upper Limit             |
|-----------------------------|-------------------------|
| $ \ddot{z} $                | 5 cm/s <sup>2</sup>     |
| $ \dot{\phi} $              | 0.2°/s                  |
| $ \dot{\theta} $            | 0.2°/s                  |
| $ \dot{\psi} $              | 0.5°/s                  |
| $ \dot{V}_{\text{vbd}} $    | 0.05 cm <sup>3</sup> /s |
| $ \dot{x}_{\text{b}} $      | 0.01 cm/s               |
| $ \dot{\delta}_{\text{r}} $ | 2°/s                    |

348 actuators' values, high vertical acceleration or high rotational velocities, as  
 349 shown in Table 2. All points which present a non-numeric value for any  
 350 signals are removed. The data are then merged once again for each cycle and  
 351 any empty cells at this stage are cleared.

### 352 3.2.2. Data Standardisation, Feature Selection and Engineering

353 For the data-driven strategies, using the data values expressed in the  
 354 physical units may result in bias towards values with much higher mean or  
 355 standard deviation values, e.g. the VBD volume. Hence, for these data-based  
 356 solutions, the input data is standardised (Goodfellow et al., 2016), as this  
 357 has been found to be more effective than normalisation for this particular  
 358 case.

359 Feature selection and engineering are specific to each method and will  
 360 thus be covered in the following individual sections.

### 361 3.3. Deep Learning for Vertical Velocity Prediction

362 Machine and deep learning strategies are increasingly being used for con-  
 363 dition monitoring applications (Fink et al., 2020). Similarly to the model-  
 364 based diagnostics, the error between the predicted and actual vertical velocity  
 365 can be used to indicate anomalies. Hence, for UGs it is possible to simplify  
 366 the anomaly detection system to the tracking of the loss of the predicted  
 367 vertical velocity in steady-state flight. The system can then be trained using  
 368 data coming from deployments representing healthy baseline conditions. By  
 369 tracking the prediction error, which is expected to rise for damaged gliders,  
 370 it is possible to identify faults. Limiting the analysis to steady-state data  
 371 greatly reduces the complexity of the problem, as simpler feedforward DNNs

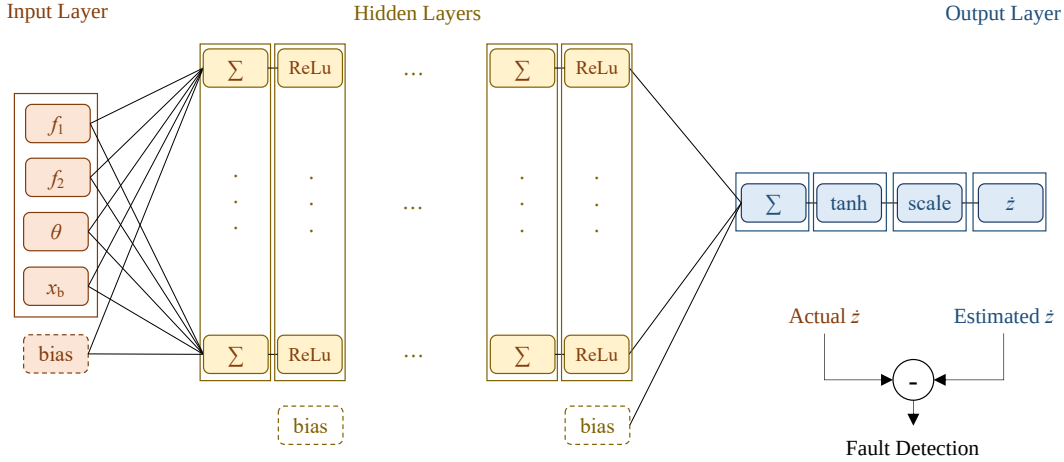


Figure 6: Diagram of the ANN used for fault detection for UGs.

372 may be used instead of more complex recurrent or convolutional architectures  
 373 (Fink et al., 2020).

374 LeCun et al. (2015) define a deep neural network as a network with more  
 375 than one hidden layer. However, other than training algorithms and activa-  
 376 tion functions, the main difference between classical machine learning and  
 377 deep learning consists in the size of the dataset analysed (Goodfellow et al.,  
 378 2016). Machine learning algorithms require significant levels of feature en-  
 379 gineering and their performance does not improve after a specific dataset  
 380 size. With deep learning solutions, even raw signals can be directly used as  
 381 features. Deep learning solutions are better scalable, thus presenting much  
 382 higher prediction accuracy than conventional machine learning strategies for  
 383 extremely large datasets (i.e. with more than approximately  $10^6$  points).

384 Here, two neural networks are considered and will be labelled as artificial  
 385 neural network (ANN) and deep neural network (DNN) despite both of them  
 386 having more than one hidden layer. Both will analyse the same cleaned data  
 387 as for the dynamic model. However, while for the ANN additional feature  
 388 engineering is performed, the DNN relies on the unmodified features.

From the dynamic model, the features of interest to predict the vertical velocity of the vehicle are identified to be the VBD volume, the pitch angle, the battery position, the water density, pressure and temperature. Hence,



the following two engineered features are obtained for the ANN:

$$f_1 = V_{\text{vbd}}\rho, \quad (5a)$$

$$f_2 = \rho V_0 (1 - \epsilon_c p + \alpha_T (T - T_0)). \quad (5b)$$

389 The first feature,  $f_1$ , is expected to be dominant. To understand whether  
 390 the second feature,  $f_2$ , should also be selected, a hypothesis test is run on the  
 391 root mean square error (RMSE) values of the test set for the ANN both with  
 392 and without  $f_2$  to assess whether the results are statistically different. Firstly,  
 393 the Lilliefors test is run to check whether the RMSE values are normally  
 394 distributed. As the error distribution does not belong to the normal family,  
 395 the non-parametric Wilcoxon signed rank test is used to assess whether the  
 396 difference of the two distributions has zero median. For both hypothesis tests,  
 397 a significance value of 5% is employed (Gibbons and Chakraborti, 2011). The  
 398 results of the tests can be found in Section 4.3.

399 Additionally, the training set of the standardised data is subdivided into  
 400 training, [holdout](#) and test sets in the following proportions: 85%, 10% and  
 401 5%. The training set contains examples used to fit the model parameters  
 402 during learning. The [holdout](#) set is employed to tune hyperparameters during  
 403 training. The test set enables an evaluation of the final, fully trained system  
 404 (Goodfellow et al., 2016).

405 The neural networks used for the fault detection are shown in Fig. 6  
 406 and Fig. 7, respectively. To detect faults, the RMSE and mean absolute  
 407 error (MAE) of the predicted vertical velocities as compared with the actual  
 408 measured  $\dot{z}$  are tracked. Both networks have a number of hidden layers with a  
 409 number of neurons that needs to be optimised. Whilst the hidden layers have  
 410 ReLu activation functions (Goodfellow et al., 2016), the output layer has a  
 411 hyperbolic tangent activation function. The Adam optimisation algorithm  
 412 is used for training (Kingma and Ba, 2015), with  $L_2$  regularisation to avoid  
 413 overfitting and using the loss (or MAE) to assess convergence (Goodfellow  
 414 et al., 2016).

### 415 3.4. Autoencoder

416 Autoencoders are a successful unsupervised learning strategy for fault de-  
 417 tection (Reddy et al., 2016; Fink et al., 2020). An autoencoder is a DNN that  
 418 comprises a dimensionality reduction step followed by a data reconstruction  
 419 step. The input and output features are thus identical for the autoencoder.  
 420 By compressing and reconstructing the data, the autoencoder is able to learn

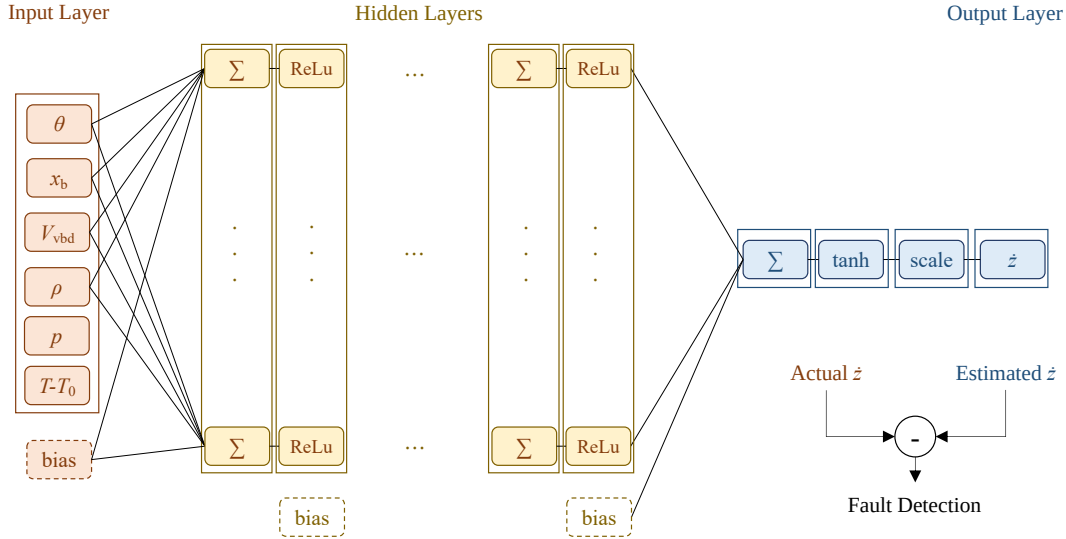


Figure 7: Diagram of the DNN used for fault detection for UGs.

421 a low-dimensional representation by identifying correlated variables and ig-  
 422 noring signal noise (Goodfellow et al., 2016). For fault detection, training  
 423 the autoencoder on a dataset of a fully operational system enables the identi-  
 424 fication of anomalies by tracking the autoencoder error in its reconstruction  
 425 of the baseline features.

426 For the anomaly detection system for UGs, the autoencoder shown in  
 427 Fig. 8 is adopted. The selected features are the vertical velocity, the product  
 428 of the VBD volume and the water density, the pitch angle and the battery  
 429 position. Adding the water temperature and pressure was found to cause high  
 430 levels of noise with little benefit on the overall prediction. Adding the roll  
 431 angle would cause the response of the autoencoder at predicting anomalies  
 432 to improve. However, the roll tilt sensor is not fundamental for the operation  
 433 of the vehicle. Hence, it has been preferred to only use signals fundamental  
 434 for the operation of the UGs in the autoencoder.

435 As only four features are used, a single hidden layer is sufficient. To  
 436 select the number of hidden neurons, a principal component analysis (PCA)  
 437 (Jolliffe and Cadima, 2016) is run. Both hidden and output neurons present  
 438 a hyperbolic activation function. The scaling layer in the output layer is  
 439 required because the output data (like the input) is standardised rather than  
 440 normalised, i.e. it goes beyond the range  $(0, 1)$ . By tracking the average

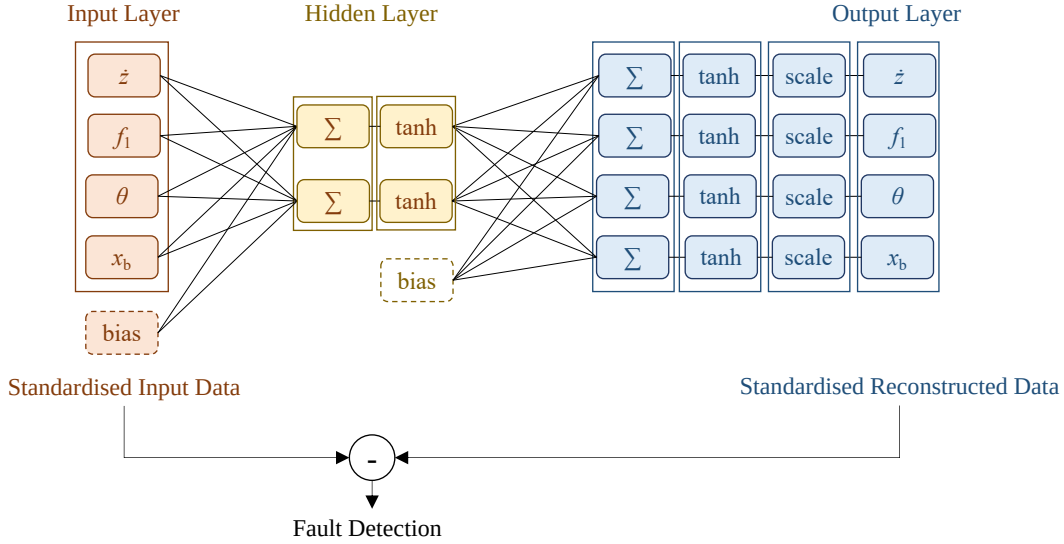


Figure 8: Diagram of the autoencoder used for fault detection for UGs.

441 RMSE of the prediction of the output features, it is possible to identify  
 442 anomalies.

443 Similarly to the DNNs, the training data set is subdivided into training,  
 444 [holdout](#) and test sets in the following proportions: 85%, 10% and 5%. The  
 445 Adam optimisation algorithm is used for training (Kingma and Ba, 2015),  
 446 with  $L_2$  regularisation to avoid overfitting and using the loss (or MAE) to as-  
 447 sess convergence (Goodfellow et al., 2016). Most often the Kullback-Leibler  
 448 divergence is employed, especially in image recognition tasks (Goodfellow  
 449 et al., 2016), but here the selected architecture has been found to be partic-  
 450 ularly effective.

#### 451 4. Results and Discussion

452 In this section, the deployment data is firstly analysed and visualised to  
 453 gather insights into the UG dynamics. From this study, the data from specific  
 454 deployments is selected for the training and testing of the deep learning  
 455 solutions. Finally, the performance of the new anomaly detection system is  
 456 discussed.

457 *4.1. Data Averaged over each Dive Cycle*

458 Figures 9-11 show the variation in the mean vertical velocity, VBD vol-  
459 ume, pitch angle, battery position, roll, and rudder angles with mission du-  
460 ration for all Slocum UGs in the study only up to 80 days after the start of  
461 each deployment. The yaw angle is not shown, as its signal is very noisy due  
462 to ocean currents.

463 From Fig. 9-11, the following observations can be made of the deploy-  
464 ments:

- 465 – Unit 345 (2014 deployment) presents a significant gap in the data,  
466 which may have been caused by retrieval and reset at sea.
- 467 – Unit 345 (2019 deployment) exhibits interesting changes in roll of small  
468 magnitude throughout the mission probably due to strong disturbances,  
469 such as transverse ocean currents.
- 470 – Unit 194 presents an abrupt change in vertical velocity, VBD volume,  
471 pitch angle and battery position for the last few dive cycles (Fig. 9-10).  
472 As the changes affect all variables in the correct direction, this effect is  
473 possibly to be due to the pilot’s decision.
- 474 – Unit 194 presents a significant angle of list of approximately  $9^\circ$ . The  
475 list angle can be identified in Fig. 11, as the roll angle has the same  
476 value in descents and ascents. Although the high roll angle may be  
477 caused by a sensor failure, which can happen on UGs (Brito et al.,  
478 2014), the corresponding rudder angle with an opposite sign indicates  
479 the effect is real. The UG is compensating for the list angle and thus  
480 asymmetric drag and lift forces with a constant rudder angle, which  
481 causes additional parasitic drag and thus energy expenditure.
- 482 – Unit 399 shows a decrease in the magnitude of the vertical velocity with  
483 time in both descents and ascents despite an increase in VBD volume  
484 in climbs. This behaviour is likely to be associated with marine growth  
485 (Haldeman et al., 2016), although no evidence of this is available from  
486 the records.
- 487 – Units 423 and 424 present much greater input VBD volume for the  
488 same mean vertical velocity than the other UGs. After obtaining the  
489 drag coefficient using the steady-state dynamic model as described in  
490 (Anderlini et al., 2020a), it is obvious that the behaviour is caused

491 by a much greater drag coefficient (approximately 0.3 as compared  
492 with 0.15). After a closer inspection, both devices are found to be  
493 ocean microstructure gliders (OMG) equipped with Rockland Scientific  
494 turbulence sensors.

495 – For units 304 and 436, wing loss is clearly recognisable as a sudden event  
496 after 57.6 and 24.5 days from the start of the deployment, respectively.  
497 In particular, from Fig. 11, it is possible to recognise that Slocum 304  
498 has lost the right wing, while Slocum 436 has lost its left wing as  
499 described in (Anderlini et al., 2020a). In Fig. 9, Slocum 436 presents  
500 further clear changes in the mean vertical velocity, VBD volume and  
501 roll angle at some point after failure, which is probably caused by the  
502 pilot to match a new mission objective. Additionally, after the loss of  
503 the wing, both units 304 and 436 either present higher upwards vertical  
504 velocity in ascents for the same VBD setting or a lower VBD volume  
505 for the same vertical velocity and vice versa in descents. In particular,  
506 the VBD needs to be fully retracted to enable the glider to dive. This  
507 behaviour is caused by the increase in positive buoyancy due to the  
508 wing loss.

#### 509 4.2. Data Selection

510 After the analysis of the averaged signals and the parameters of the dy-  
511 namic model, the deployments have been subdivided into training and vali-  
512 dation datasets as shown in Table 3. The table also summarises the status of  
513 the UG during the deployment. Hence, it is clear that only healthy baseline  
514 deployments have been used in the training set, with the test subset used to  
515 assess the methods’ accuracy on the baseline data. The validation datasets  
516 are used to assess the ability of the anomaly detection schemes to detect  
517 previously unseen anomalies on new real UG deployments..

518 Additionally, Fig. 12 shows the subdivision of the dataset into each cate-  
519 gory of glider status for the cleaned data points. As can be seen, the dataset  
520 is unbalanced and skewed towards healthy baseline operations. This is common  
521 for condition monitoring problems, where engineering systems are designed  
522 to have high reliability (Fink et al., 2020). As the anomaly detection system  
523 presented here is based on unsupervised learning solutions that are trained  
524 only from healthy baseline gliders, the methods are not negatively impacted  
525 by the imbalance. Conversely, the limited data corresponding to anomalous

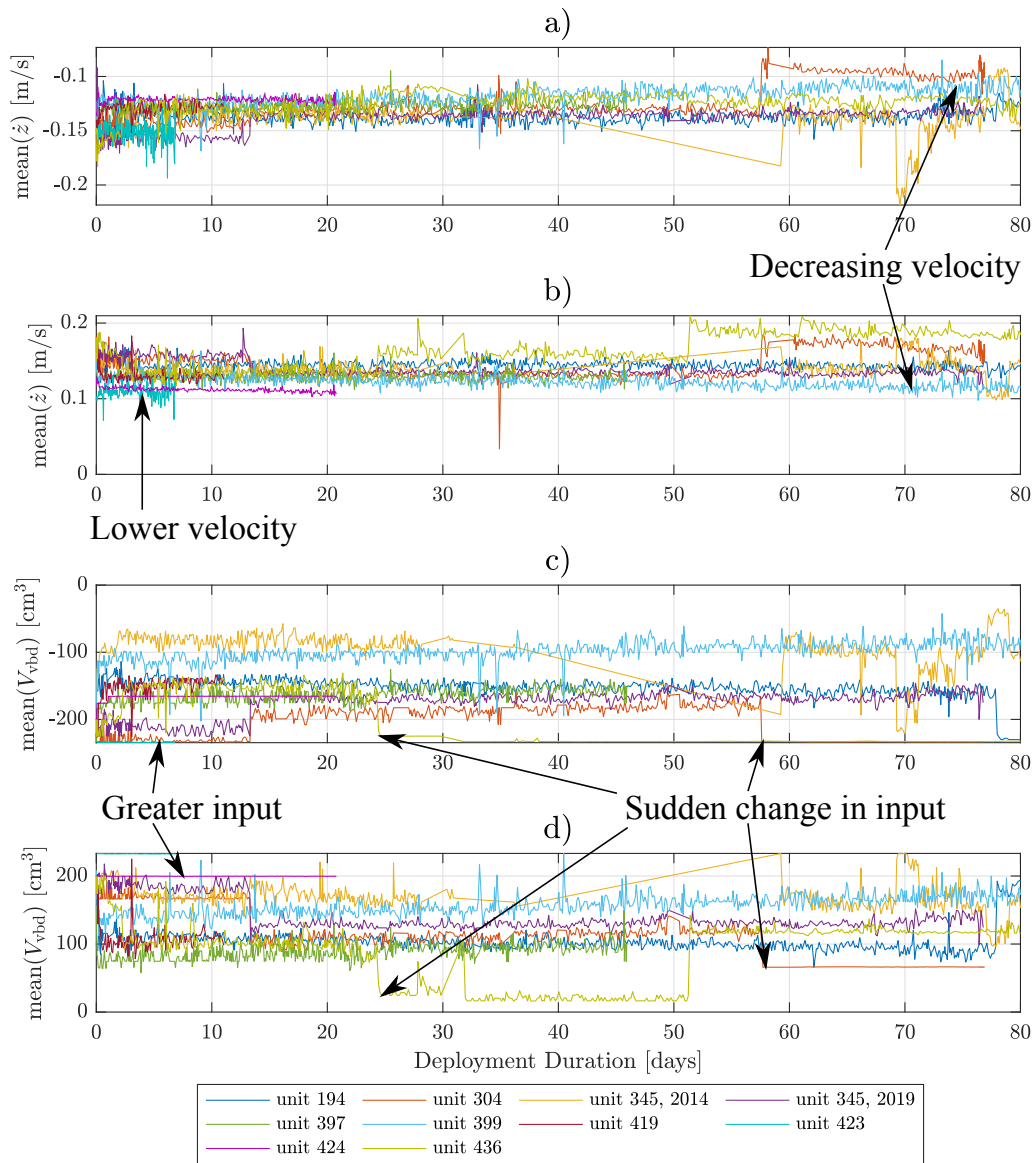


Figure 9: Mean vertical velocity and VBD volume in dives (a,c) and climbs (b,d), respectively.

526 behaviour will pose significant challenges for future fault diagnostics studies  
 527 relying on a classification task.

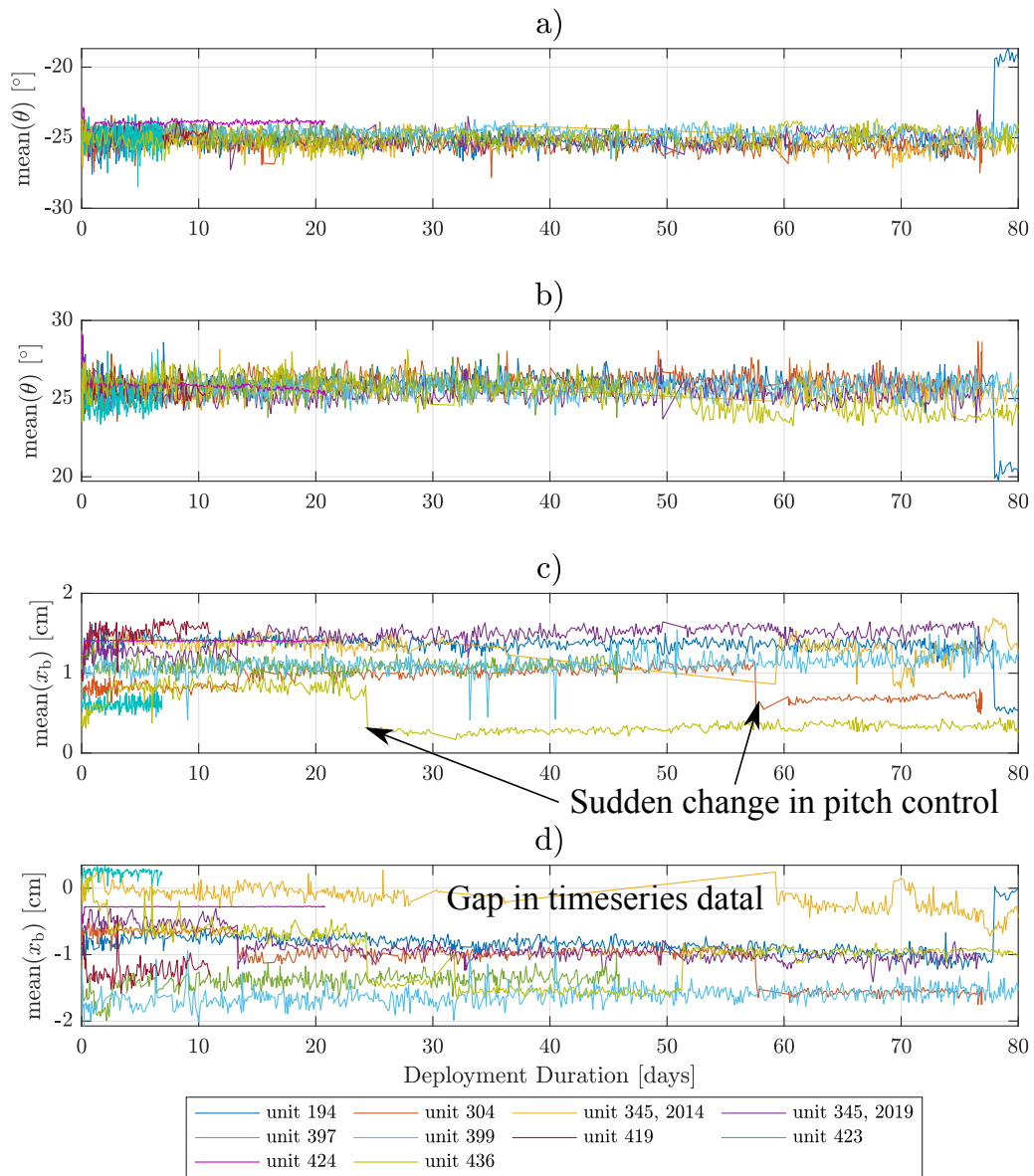


Figure 10: Mean pitch angle and battery position in dives (a,c) and climbs (b,d), respectively.

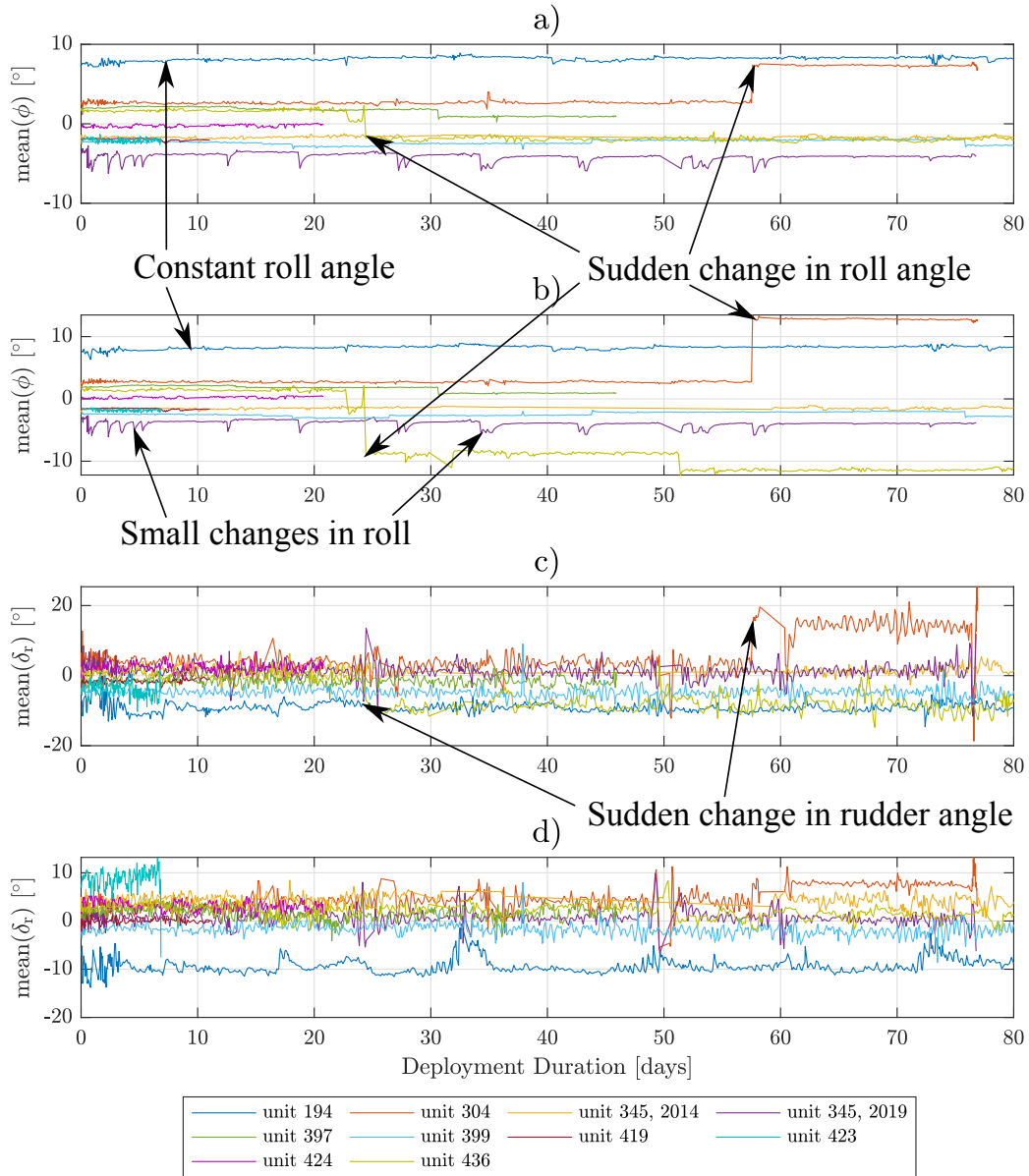


Figure 11: Mean roll and rudder angles in dives (a,c) and climbs (b,d), respectively.

528 *4.3. Anomaly Detection Results*

529 From hyperparameter optimisation, three hidden layers with eight neu-  
 530 rons each are selected for the ANN and four hidden layers with twelve neurons



Table 3: Subdivision of the deployments into training (further subdivided into training, holdout and test subsets) and validation data sets.

| No. | Unit | Status              | Dataset    | No. points |
|-----|------|---------------------|------------|------------|
| 1   | 345  | healthy             | training   | 380,236    |
| 2   | 345  | strong disturbances | training   | 325,279    |
| 3   | 397  | healthy             | training   | 287,139    |
| 4   | 419  | healthy             | training   | 68,615     |
| 5   | 194  | angle of list       | validation | 492,585    |
| 6   | 399  | possible biofouling | validation | 592,680    |
| 7   | 423  | OMG                 | validation | 22,400     |
| 8   | 424  | OMG                 | validation | 122,366    |
| 9   | 304  | wing loss           | validation | 275,103    |
| 10  | 436  | wing loss           | validation | 555,774    |

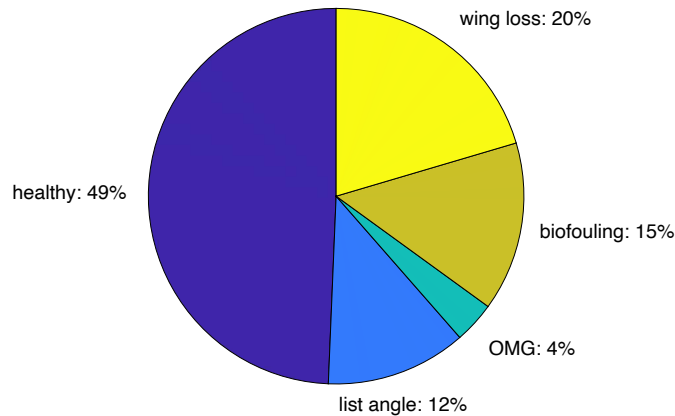


Figure 12: Share of the cleaned dataset corresponding to different types of glider behaviour. The total number of points after cleaning is 4,067,957.

531 each for the DNN. The PCA returned percentage total variance of 95.59%,  
 532 3.25%, 0.93% and 0.24% for the first four principal components for healthy  
 533 baseline gliders. Hence, two neurons are selected for the hidden layer of the  
 534 autoencoder.

535 For the ANN, the Wilcoxon signed rank test on the RMSE in the test  
 536 set with and without  $f_2$  presents a probability of  $5.3 \times 10^{-7} \ll 0.05$ . Hence,  
 537 adding the feature results in a significant change. Similarly, the Lilliefors  
 538 test results in a probability of  $0.001 \ll 0.05$ , so that the data is not normally

539 distributed.

540 The algorithms are implemented in MATLAB using the Deep Learning  
541 toolbox, selecting the default settings for training. Due to the large dataset,  
542 a minimum batch size of 512 were selected and four epochs were found to  
543 be sufficient for convergence for the ANN and DNN (corresponding to ap-  
544 proximately 10,400 steps) and five for the autoencoder (corresponding to  
545 approximately 12,800 steps).

546 Fig. 13 shows the time variation of the RMSE of the trained deep learning  
547 solutions for the six test datasets selected in Table 3. [In this work, the RMSE](#)  
548 [of the data-driven methods represents the anomaly score that can be used](#)  
549 [to inform classification algorithms for fault diagnostics in the future; hence,](#)  
550 [accuracy values will not be presented.](#) Additionally, for units 304 and 436  
551 the rule- and model-based measures to detect wing loss from Anderlini et al.  
552 (2020a) are included, namely the difference in the mean roll angle in ascents  
553 and descents and the buoyancy offset, respectively. For unit 399, the drag  
554 coefficient is the model-based metric to detect marine growth as in Anderlini  
555 et al. (2020b).

556 As can be seen in Fig. 13, all methods are effective in detecting wing loss  
557 for units 304 and 436. In particular, all schemes present a clear separation  
558 from the initial healthy baseline conditions and a sudden increase in the  
559 anomaly detection metric after the event, even though the RMSE of the  
560 autoencoder presents a more clearly defined change than that of the ANN and  
561 DNN. For the initial dives, the anomaly detection metrics present significant  
562 oscillations, as these are shallow, highly dynamic dives.

563 For unit 399, the autoencoder RMSE and the drag coefficient are the  
564 clearest indicators of growing levels of biofouling. Conversely, the RMSE of  
565 the ANN and DNN do not grow significantly.

566 [The mean and standard deviation of the anomaly score for the three](#)  
567 [methods for the different anomalies can be seen in Table 4.](#) The wing loss  
568 [data in the table has been split before and after the event.](#) Note that the  
569 [absolute magnitude of the anomaly score is not important and varies with the](#)  
570 [method; conversely, the relative change in anomaly score between anomalous](#)  
571 [UGs and the healthy baseline is critical.](#)

572 For units 194 (angle of list) and 424 (OMG), [the mean anomaly score is](#)  
573 [not dissimilar from the mean values for the baseline value.](#) This indicates  
574 that without the roll sensor input, the data-driven methods struggle to detect  
575 an angle of list, as it does not negatively impact the dynamics of the UG.  
576 Additionally, while the methods are able to generalise well to higher drag

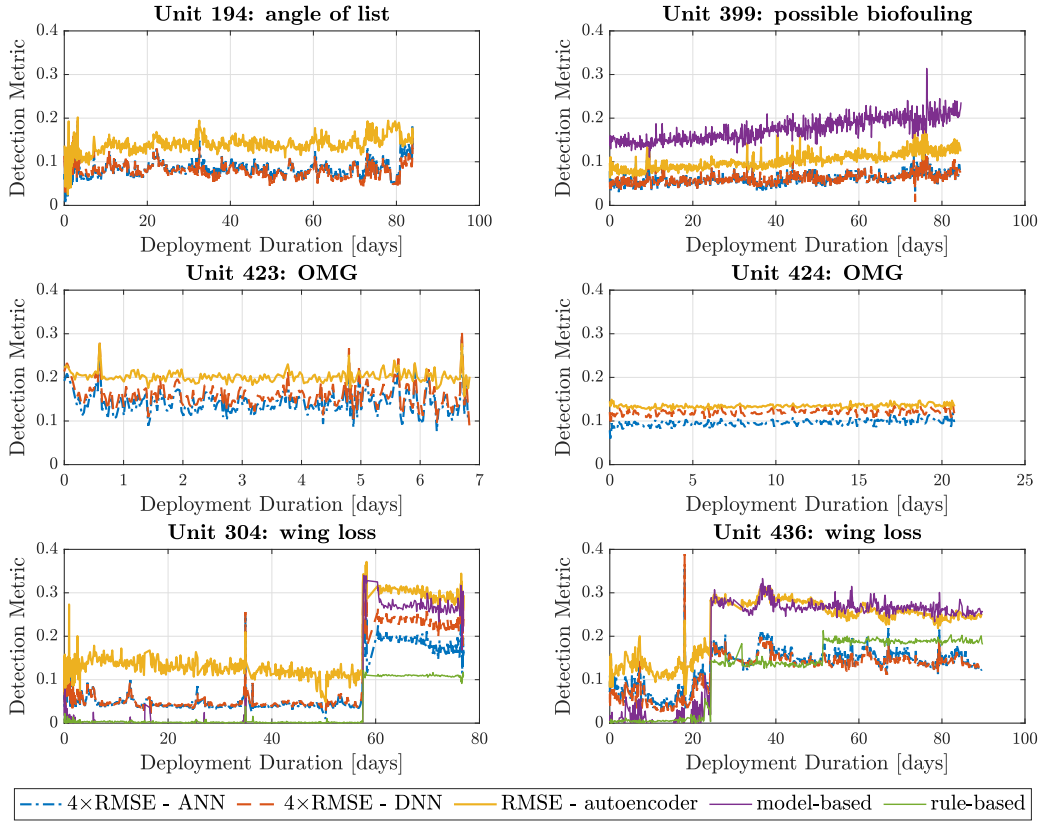


Figure 13: Anomaly detection metrics from the ANN, DNN and autoencoder for the six verification deployments. The model-based metric is the buoyancy offset (divided by 4) [N] and the rule-based metric the difference between the mean roll angle in ascents and descents (divided by 50) [ $^{\circ}$ ] for units 304 and 436 and the drag coefficient for unit 399.

577 levels caused by large appendages for the OMG unit 424, the anomaly detec-  
 578 tion metric are higher by 30%-50% than for the baseline for unit 423. The  
 579 much shorter deployment for unit 423 hints at possible problems, but further  
 580 investigation is required to confirm the ability of the methods to generalise  
 581 to different platforms.

582 It is clear that the autoencoder and deep learning methods are more  
 583 general than the bespoke rule- or model-based heuristics and can help detect  
 584 a good range of anomalies. In particular, the dynamic model is specific  
 585 to Slocum UGs, especially the model for the compressibility and thermal  
 586 expansivity of its hull, and even differs from other UGs, like Seaglider UGs

Table 4: Mean ( $\mu$ ) and standard deviation ( $\sigma$ ) of the anomaly score for each data-driven method for the analysed subsets. Note that only the test set has been used for the training data. The wing loss data has been split before and after the event.

| Unit          | Status        | $\mu_{\text{ANN}}$ | $\sigma_{\text{ANN}}$ | $\mu_{\text{DNN}}$ | $\sigma_{\text{DNN}}$ | $\mu_{\text{auto}}$ | $\sigma_{\text{auto}}$ |
|---------------|---------------|--------------------|-----------------------|--------------------|-----------------------|---------------------|------------------------|
| 345, 397, 419 | healthy       | 0.0137             | 0.0127                | 0.0131             | 0.0120                | 0.0978              | 0.0422                 |
| 194           | angle of list | 0.0210             | 0.0047                | 0.0201             | 0.0045                | 0.1399              | 0.0193                 |
| 399           | biofouling    | 0.0154             | 0.0035                | 0.0153             | 0.0033                | 0.1031              | 0.0185                 |
| 423           | OMG           | 0.0358             | 0.0075                | 0.0401             | 0.0075                | 0.2020              | 0.0138                 |
| 424           | OMG           | 0.0241             | 0.0020                | 0.0302             | 0.0018                | 0.1347              | 0.0045                 |
| 304           | healthy       | 0.0119             | 0.0041                | 0.0127             | 0.0049                | 0.1274              | 0.0229                 |
| 304           | wing loss     | 0.0465             | 0.0055                | 0.0589             | 0.0050                | 0.2981              | 0.0194                 |
| 436           | healthy       | 0.0191             | 0.0081                | 0.0160             | 0.0083                | 0.1334              | 0.0263                 |
| 436           | wing loss     | 0.0384             | 0.0049                | 0.0364             | 0.0040                | 0.2630              | 0.0205                 |

587 as described in Anderlini et al. (2019a), let alone other MAS technologies.  
 588 Hence, the autoencoder is the most promising method for the fault detection  
 589 of UGs thanks to its generality and superior performance in detecting wing  
 590 loss clearly over the ANN and DNN.

591 However, the detection of some anomalies, like the angle of list, would  
 592 need the inclusion of additional sensory or virtual signals, e.g. the angle of  
 593 roll. As the roll sensor is non-critical for the operation of the UG, the roll  
 594 signal is seldom sent to shore via satellite from operators. Hence, changes in  
 595 operational practices may be required for a fuller anomaly detection system.  
 596 Finally, the similar performance of the ANN and DNN methods indicates  
 597 that high effort in feature engineering is not necessary when the dataset is  
 598 large.

## 599 5. Conclusions

600 Increasing the use of MAS over the horizon requires on-line remote anomaly  
 601 detection, since adverse behaviour may result in the loss of the vehicle or rep-  
 602 resent risk to other sea users. Currently, vehicles are monitored by expert  
 603 pilots, thus limiting the scale of fleets of MAS that may be deployed simulta-  
 604 neously. In this study, different data-driven solutions including feedforward  
 605 DNN and autoencoders have been introduced that are able to successfully  
 606 detect a range of anomalies, including wing loss and marine growth. The  
 607 data-driven methods are more general than physics-induced bespoke heuris-  
 608 tics of rule- and model-based solutions, and are able to generalise to different

609 sensor configurations for UGs, including OMG. The autoencoder RMSE pro-  
610 vided the clearest indicator for wing loss, while the change in the ANN and  
611 DNN RMSE after the sudden event was detectable, but not as clear.

612 In this work the fault detection system was developed off-line using data  
613 collected over past field experiments. The system is being incorporated  
614 within the NOC’s command and control architecture for MAS over the hori-  
615 zon. Hence, it will provide remote, on-line condition monitoring of MAS  
616 during future deployments. The inclusion of additional signals and the con-  
617 sideration of dynamic effects will be key to extend the system to the detection  
618 of additional anomalies, e.g. angle of list and sensor loss. **Field tests will need**  
619 **to be undertaken to collect data for additional anomalies to fully validate the**  
620 **efficacy and generality of the data-driven solutions.** Furthermore, anomaly  
621 detection will need to be extended into full fault diagnostics to identify the  
622 affected subsystems.

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