Thermal filters for the ATHENA XIFU: ongoing activities towards the conceptual design

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Abstract ATHENA is the L2 mission selected by ESA to pursue the science theme "Hot and Energetic Universe". One of the two instruments at the telescope focal plane is the X-ray Integral Field Unit, an array of TES microcalorimeters operated at T < 100 mK. To allow the x-ray photons focused by the telescope to reach the detector, windows have to be opened on the cryostat thermal shields. X-ray transparent filters need to be mounted on these open windows to attenuate the IR radiation from warm surfaces, to attenuate RF electromagnetic interferences on TES sensors and SQUID electronics, and to protect the detector from contamination. This paper reviews the ongoing activities driving the design of the XIFU thermal filters.

Keywords X-ray Astronomy • ATHENA • X-IFU • Thermal Filters

1 Introduction

ATHENA[1] is an X-ray observatory, designed to address the "Hot and Energetic Universe" science theme[2], selected by ESA for L2 – the second Large-class mission within the Cosmic Vision (launch scheduled in 2028). One of the two instruments at the focal plane of the large area telescope[3] is the X-ray Integral Field Unit (XIFU), an array of transition edge sensor (TES) microcalorimeters operating at temperatures < 100 mK, providing

high spectral resolution ($\Delta E_{FWHM} = 2.5 \text{ eV}$ at E < 7 keV) and imaging in the energy range 0.2-12 keV, over a 5' diameter field of view[4].

To allow the x-ray photons focused by the telescope to reach the detector, windows have to be opened on the cryostat shields. X-ray transparent filters need to be mounted on such shields to reduce IR radiation heat load and to prevent degradation of the energy resolution by photon shot noise[5]. The development of the thermal filters (TF) contains many technical challenges: small thickness for high transparency at soft X-rays, large size to cover the FOV, complex interface with the cryostat and focal plane assembly, detector protection from contamination, EMI attenuation at the telemetry RF X-band. For these reasons the design, verification, and calibration of the TF are critical tasks in the ATHENA XIFU implementation.

2 Filter Design and Performance Modeling

Previous space experiments with doped-silicon X-ray micro-calorimeters, have used TF based on aluminum coated polyimide[6-10]. The ATHENA XIFU has similar requirements on attenuation of IR radiation, however, the use of TES sensor technology and SQUID based electronics put an additional requirement on shielding RF EMI from the spacecraft operation and telemetry. Based on such experience and new requirements on RF shielding, we baseline polyimide and aluminum for all TF with metal meshes on the two larger diameter filters. Fig. 1 *left panel* shows the modeled UV-VIS-IR transmission[5] for a set of five filters with a total of 225 nm polyimide and 150 nm aluminum. Upon our experience[11,12], we have assumed that 5 nm of oxide build up on each side of the Al coating and become transparent in the UV/Vis. Fig. 1 *right panel* shows the modeled X-ray transmission for the same set of filters.

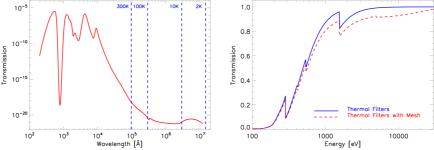


Fig. 1 UV/VIS/IR transmission (*left panel*) for a set of five TF with a total of 225 nm polyimide and 150 nm Al. X-ray transmission (*right panel*) for the same set of filters without meshes (*solid line*) and with a 96% open area Ti mesh on the two larger diameter filters (*dashed line*). Color figure online

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2.1 Thermal Modeling

Fig. 2 shows the TF configuration inside the cryostat adopted for the thermal modeling performed with Comsol Multiphysics®. In the simulation we have presently taken into account the aluminum layer of each filter and the metal meshes in the two larger filters, while we have neglected the thin polyimide films due to their much lower thermal conductivity and specific heat. Table 1 shows the derived equilibrium temperatures at the center of the five filters.

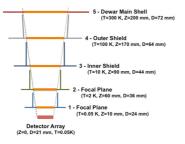


Fig. 2 Schematic TF configuration adopted in the thermal modeling. Color figure online.

Table 1 Temperature of the cryostat shields and modeled equilibrium temperature at the center of the five thermal filters.

	Shield T [K]	Filter T [K]
F1	0.05	0.87
F2	2	2.1
F3	10	10.4
F4	100	183
F5	300	255

2.2 Radiation Heat Load and Photon Shot Noise

Adopting the temperature profile derived from the thermal modeling and assuming that filters are tilted by 1° in alternate directions to reduce multiple reflections, we have calculated the total radiation power onto the detector array and the energy resolution degradation due to photon shot noise (Table 2), according to the recipe in Barbera et al. 2014[5]. Notice that if the filters are not tilted, the multiple reflections significantly increases the radiation power onto the detector and the energy resolution increases to > 10 eV.

Table 2: Radiation power and energy resolution degradation due to photon shot noise from each of the five filters. For each filter we also report: central equilibrium temperature (T), distance from the array (Z), and diameter (D).

	T	Z	D	Power	ΔE_{FWHM}
	[K]	[mm]	[mm]	[pW]	[eV]
F1	0.87	10	24	8033	0.195
F2	2.1	60	36	4.446	0.007
F3	10.4	90	44	0.286	0.004
F4	183	170	64	7.225	0.128
F5	255	200	72	0.092	0.016
Total				8045	0.234

2.3 Structural Analysis

Since the vibration levels of the XIFU FPA and cryostat aperture cylinder are not yet known, we performed a preliminary structural analysis under 10

mbar static pressure (qualification static load for the XMM-Newton EPIC filters) on the mechanically more challenging larger filter (D=72 mm) where we foresee the use of a reinforcing mesh. Different parameters, namely: mesh material, geometry, pitch (P), bar width (BW), and thickness (T) have been investigated. The currently most promising mesh consists of two Ti alloy Ti6Al4V layers with honeycomb geometry: 1) a structural layer with P=5 mm, BW=50 μ m, T=250 μ m, and 2) an RF attenuator layer with P=1 mm, BW=10 μ m, T=30 μ m. The blocking factor of such mesh is nearly 4%. The calculated maximum tensile stress on the structural mesh and RF attenuator are 560 MPa and 470 MPa, respectively, both values well within the ultimate tensile strength of the material (950 MPa). The maximum tensile stress on a polyimide film 45 nm thick attached to such mesh is 140 MPa also well within the ultimate tensile strength of the material (310 MPa).

2.4 Radio Frequency EMI attenuation

The telemetry uplink/downlink signals entering the instrument FOV by direct illumination or diffraction from satellite and instrument structures can cause EMI on TES detectors and SQUID electronics. A preliminary Electro Magnetic FEM was implemented with a 60 mm length standard WR-90 X-band waveguide, using the HFSS software (Ansys Corporation), to evaluate the attenuation from conductive meshes in the telemetry X-band frequency range [8.2÷12.4 GHz]. The mesh geometry (e.g. honeycomb, square, radial) is irrelevant to first order with respect to parameters such as mesh pitch, bar width, and thickness. Fig. 3 shows the electric field attenuation vs. frequency for a 20 μ m thick mesh with different pitches and bar widths. The use of a mesh with 1 mm pitch, and 10 μ m bar width on the two larger diameter TF should provide four orders of magnitude attenuation in RF power at 10 GHz.

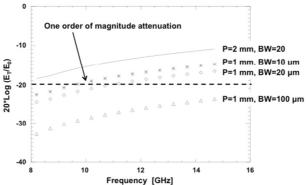


Fig. 3 Electric field attenuation vs. frequency for a 20 μ m thick mesh with different pitches (P) and bar widths (BW).

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3 Optical Load from Astrophysical Sources

The optical load from the bright UV/VIS counterparts of x-ray sources such as massive stars, AGN's in outburst, Mars, etc., can degrade the XIFU energy resolution by photon shot noise. Fig. 4 (solid lines) shows the calculated XIFU energy resolution degradation vs. visual magnitude for massive stars in the $T_{\rm eff}$ range 15000÷45000 K, assuming UV/VIS stellar spectra attenuation only by the TF. Even the faintest hot stars would have a non negligible degradation of the spectral resolution ($\Delta E_{\rm FWHM} > 0.2~{\rm eV}$). On the other hand, the XIFU capability to observe hot stars as bright as $m_{\rm v} = 2~{\rm is}$ mandatory to probe the dynamics of stellar winds in O-B stars [13]. Fig. 4 (dashed lines) shows the same calculation with the use of two optional optical blocking filters (OBF) in the filter wheel (FW): a thin OBF (200 nm polyimide + 40 nm Al) to observe hot stars fainter than $m_{\rm V} \sim 7.5$ with no significant degradation of energy resolution, and a thick OBF (200 nm polyimide + 80 nm Al) to observe hot stars as bright as $m_{\rm v} = 2$.

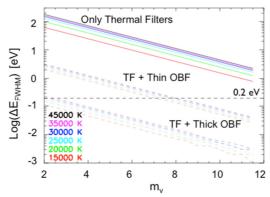


Fig. 4 XIFU energy resolution degradation due to optical load from hot stars. The *solid lines* include UV/VIS stellar spectra attenuation by the TF, while the *dashed lines* include the attenuation by additional OBF. Color figure online.

4 Summary and Conclusions

Thermal filters (TF) mounted on the cryostat aperture cylinder and FPA are needed to reduce the IR radiation onto the XIFU detector array, to maintain the photon shot noise well below the nominal energy resolution, and to attenuate RF EMI onto the sensitive TES and SQUID read-out electronics.

Based on the heritage from previous missions we have chosen polyimide films coated with aluminum as the baseline materials for the TF. The currently investigated design consists of 5 separate filters, operating at different temperatures, for a total of 225 nm polyimide and 150 nm aluminum. Such configuration provides adequate attenuation of IR radiation.

We plan to use metal meshes (open area > 95%) for the two larger diameter filters to provide mechanical reinforcement, EMI RF attenuation, and, if needed, capability to warm up the filters for de-contamination. Currently investigated mesh consists of two layers of Ti alloy Ti6Al4V: 1) a structural layer with 5 mm pitch, 50 μ m bar width, 250 μ m thickness, and 2) an RF attenuator with 1 mm pitch, 10 μ m bar width, 30 μ m thickness. Preliminary structural analysis under a 10 mbar static load show a maximum tensile stress both on the mesh and on the attached polyimide film well within the ultimate tensile strengths of the adopted materials.

The optical load from the bright UV/VIS counterparts of x-ray sources (e.g. massive stars, AGN's, Mars) can degrade the XIFU energy resolution by photon shot noise. We propose to use two optional optical blocking filters (OBF) on the filter wheel, namely: a thin OBF (200 nm polyimide + 40 nm Al) to observe hot stars fainter than $m_V \sim 7.5$, and a thick OBF (200 nm polyimide + 80 nm Al) to observe hot stars as bright as $m_V = 2$.

A first set of filters will be procured in Q4 2015 to improve the TRL in the pre-assessment study phase. UV/VIS/IR transmission will be measured at different temperatures in the range 10-300 K to check the model. High spectral resolution x-ray transmission measurements and x-ray photoelectron spectroscopy will be performed to model transmission near the absorption edges in the temperature range 80-300 K and to measure the surface aluminum oxide. Environmental tests will start in Q1-Q2, 2016.

Acknowledgements We acknowledge support by the Italian Space Agency (Contract number 2014-045-R.O.) and fruitful discussions and suggestions by LUXEL corp.

References

- 1. X. Barcons et al., J. Phys. Conf. Series, 610(1), 012008 (2015).
- 2. K. Nandra et al., eprint arXiv:1306.2307N (2013).
- 3. R. Willingale et al., eprint arXiv:1307.1709W (2013).
- 4. L. Ravera et al., Proc. SPIE, 9144, 9144-2L (2014).
- 5. M. Barbera et al., Proc. SPIE, 9144, 9144-5U (2014).
- 6. D. McCammon et al., JLTP, 151, Issue 3-4, 715-720 (2008).
- 7. M.D. Audley et al., Proc. SPIE, 3765, 751-761 (1999).
- 8. R.L. Kelley et al., PASJ, 59, 77-112 (2007).
- 9. C.P. de Vries et al., Proc. SPIE, 7732, 7732-13 (2010).
- 10. T. Takahashi et al., Proc. SPIE, 7732, 7732-0Z (2010).
- 11. M. Barbera et al., Proc. SPIE, 4851, 264-269 (2003).
- 12. M. Barbera et al., Proc. SPIE, 8859, 8859-14 (2013).
- 13. S. Sciortino et al., eprint arXiv:1306.2333 (2013).